

## ER-FINS-98 High Performance FOG INS (Replace PHINS )

### Introduction

ER-FINS-98 FOG inertial navigation system (Substitute of PHINS) uses triaxial fiber optic gyro to sense angular motion and output digital signal in proportion to the carrier movement angular rate; uses three orthogonal collocation of quartz accelerometer to sense carrier linear acceleration and output proportionable current signal, which is switched to frequency signal though I-F conversion circuit, then input to navigation computer. Computer finishes data receipt of gyro, accelerometer and external GPS, system compensation error calculation, navigation solution and externally sending real-time speed, position, attitude and navigation information affixed the cycle through the monitoring port .

The initial alignment of ER-FINS-98 is divided into two modes: static alignment and double position alignment. The position accuracy of the two position alignment is higher than that of the static alignment.

ER-FINS-98 has high precision and best stability and showed excellent performance in comparison test with PHINS related INS products.

### Specifications

Pure Inertial Mode	
Azimuth accuracy	$\leq 0.1^\circ \text{sec} \psi$ ( $1\sigma$ )
Attitude accuracy	$\leq 0.02^\circ$ ( $1\sigma$ )
Azimuth holding accuracy	$0.05^\circ/\text{h}$
Attitude holding accuracy	$0.03^\circ/\text{h}$
Positioning accuracy (50%CEP)	$\leq 2\text{nm}/\text{h}$ (10min Static alignment)
Horizontal velocity precision (RMS)	$\leq 2\text{m}/\text{s}$ (10min Static alignment)
Positioning accuracy (50%CEP)	$\leq 1\text{nm}/\text{h}$ (Two position alignment, Alignment time is less than 30min)

Horizontal velocity precision (RMS)	≤ 1m/s (Two position alignment, Alignment time is less than 30min)
<b>GNSS Assisted Navigation Mode (External receiver)</b>	
Azimuth accuracy	≤ 0.1°sec $\psi$ (1 $\sigma$ )
Horizontal attitude accuracy	≤ 0.02° (1 $\sigma$ )
Azimuth holding accuracy	≤0.05°
Horizontal attitude holding accuracy	≤ 0.01° (1 $\sigma$ )
Positioning accuracy	≤ 5m (1 $\sigma$ )
Speed accuracy	≤ 0.1m/s (1 $\sigma$ )
<b>Power and environment</b>	
Data measurement frequency	Maximum 100Hz
Power Supply	23~31V DC Power Supply, Nominal Supply Voltage27V
Power	Normal temperature steady-state power consumption is less than 17W
	High and low temperature steady-state power consumption is less than 20W
	Start transient power consumption is less than 50W
Working Temp	-40°C~+60°C
Storage Temp	-45°C~+80°C
Installment Dimension	180mm×180mm×160mm
Quality	< 6kg
<b>Fiber Optic Gyroscope</b>	
Time to Prepare	≤15s
Bias Stability (100s)	≤0.02°/h (1 $\sigma$ )
Bias repeatability	≤0.02°/h (1 $\sigma$ )
Random Walk Coefficient	≤ 0.005°/√ Hz
Scale Factor Non-linearity	≤ 50ppm (1 $\sigma$ )
Scale Factor Repeatability	≤ 50ppm (1 $\sigma$ )

Gyro Measurement Range	$\geq \pm 300^\circ/\text{s}$
<b>Quartz Accelerometer</b>	
Measurement Range	1-20g~+20g
The Threshold Value	$\leq 5 \times 10^{-6}\text{g}$
Scale Factor Repeatability	$\leq 3.5 \times 10^{-5} (1\sigma)$
Scale Factor Temperature Coefficient	$\leq 6 \times 10^{-5}/^\circ\text{C} (-40^\circ\text{C} \sim +60^\circ\text{C})$
The Second Order Nonlinear Coefficient	$\leq 3 \times 10^{-5}\text{g}/\text{g}^2$
Bias	$\leq 6 \times 10^{-3}\text{g}$
Bias Repeatability	$\leq 2.5 \times 10^{-5}\text{g} (1\sigma)$
Bias Temp Coefficient	$\leq 2.5 \times 10^{-5}\text{g}/^\circ\text{C} (-40^\circ\text{C} \sim +60^\circ\text{C})$
Band width	$\geq 800\text{Hz}$